

Numerical investigation of pressure recovery in an induced draught air-cooled condenser for CSP application

Gerhard Bekker

Solar Thermal Energy Research Group
Stellenbosch University

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Background

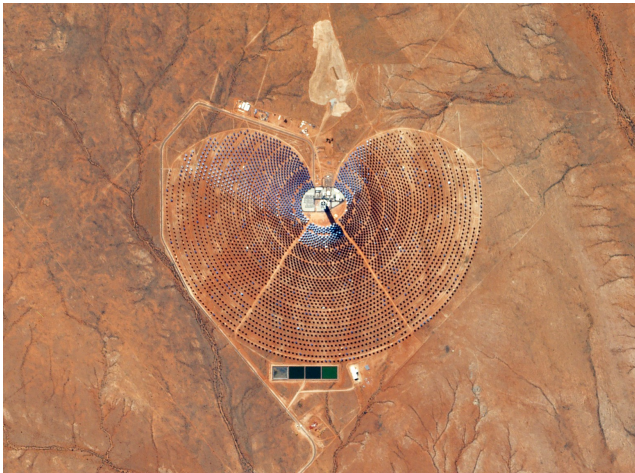


Figure: Khi Solar One

Background

Limited water resources



Figure: Kathu Solar Park

Background

Dry-cooling

Dry-cooling systems are typically employed:

Background

Dry-cooling

Dry-cooling systems are typically employed:

- Natural draught



Figure: Natural draught

Background

Dry-cooling

Dry-cooling systems are typically employed:

- Natural draught
- Mechanical draught
 - Forced draught
 - Induced draught



Figure: Mechanical draught



Figure: Natural draught

Background

Dry-cooling

Dry-cooling systems are typically employed:

- Natural draught
- Mechanical draught
 - Forced draught
 - **Induced draught**



Figure: Mechanical draught



Figure: Natural draught

Problem statement

- Outlet kinetic energy lost to atmosphere
- This is a system loss
- Decreases the total-to-static efficiency of the fan



Figure: Induced draught air-cooled condensers (ACCs)

Objective

- Minimise the outlet kinetic energy loss of the M-fan

Objective

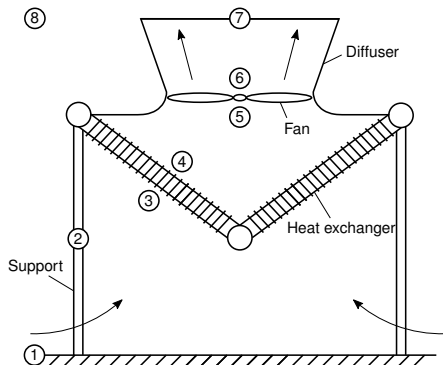
- Minimise the outlet kinetic energy loss of the M-fan
- The M-fan was designed by Wilkinson *et al.* (2017) for CSP application

Table: M-fan specifications

Diameter	24 ft (7.3152 m)
Number of blades	8
Hub-tip-ratio	0.29
Rotational speed	151 rpm
Flow rate	333 m ³ /s
Fan static pressure	116.7 Pa

Draught equation

Draught equation:
Energy supplied = energy
dissipated

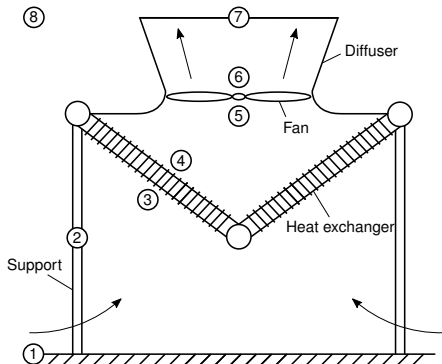


Draught equation

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Energy supplied = energy dissipated

- Dimensionless pressure loss/gain coefficient:

$$K = \frac{\Delta p}{\rho v^2 / 2}$$



Draught equation

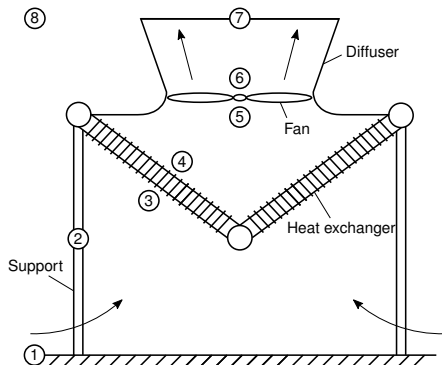
Draught equation:
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- Induced draught ACC:

$$\Delta p_{Fs} + \alpha_{eF} \rho v_{FC}^2 / 2 = \Delta p_{sys} + K_{dif} \rho v_{FC}^2 / 2 + \alpha_{e7} \rho v_7^2 / 2$$



Pressure recovery

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- Pressure recovery term:

$$K_{rec} = \alpha_{eF} - \alpha_{e7} (A_{FC} / A_7)^2 - K_{dif}$$

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$$\Delta p_{Fs} + K_{rec} \rho v_{FC}^2 / 2 = \Delta p_{sys}$$

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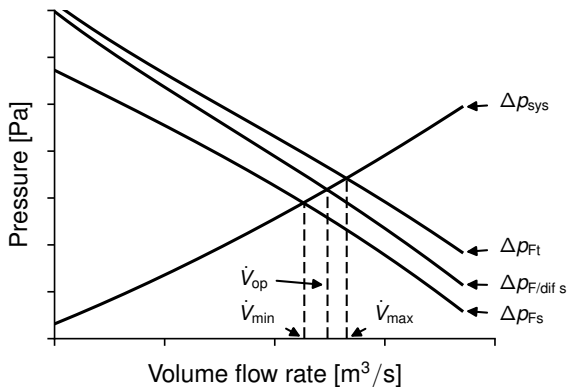
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Pressure recovery term:

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Kinetic energy factor

Pressure recovery

Kinetic energy factor: ratio of actual to mean kinetic energy through a section

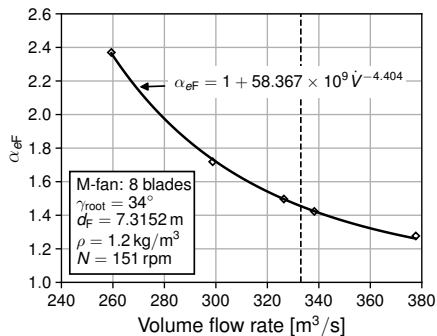
$$\begin{aligned}\alpha_{eF} &= \frac{1}{V_{FC}^3 A_{FC}} \int_A c_x (c_x^2 + c_\theta^2 + c_r^2) dA \\ &= \alpha_{eF_x} + \alpha_{eF_\theta} + \alpha_{eF_r}\end{aligned}$$

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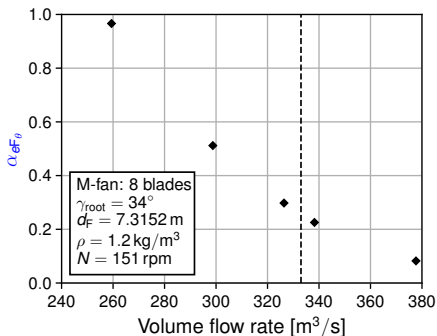
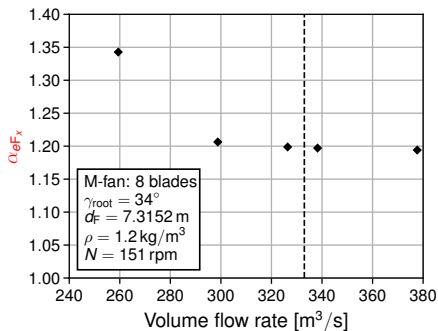


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Area ratio

Pressure recovery

Pressure recovery term:

$$K_{\text{rec}} = \alpha_{eF} - \alpha_{e7}(A_{\text{FC}}/A_7)^2 - K_{\text{dif}}$$

- Want A_7 as large as possible
- Avoid flow separation

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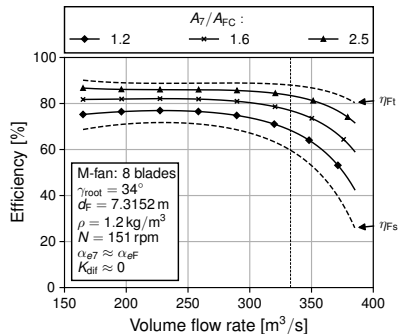


Figure: Efficiency characteristic

Diffuser loss coefficient

Pressure recovery

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- K_{dif} represents the total pressure loss across the diffuser
- Owing to viscous effects
- Aim to keep as small as possible

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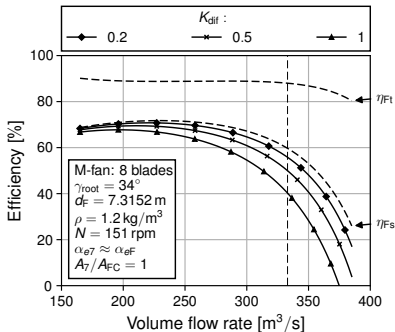


Figure: Efficiency characteristic

Validation case: ERCOFTAC conical diffuser

- Experiments performed by Clausen *et al.* (1993)
- Swirling flow in a conical diffuser
 - Total divergence angle: 20°
 - Area ratio: 2.84

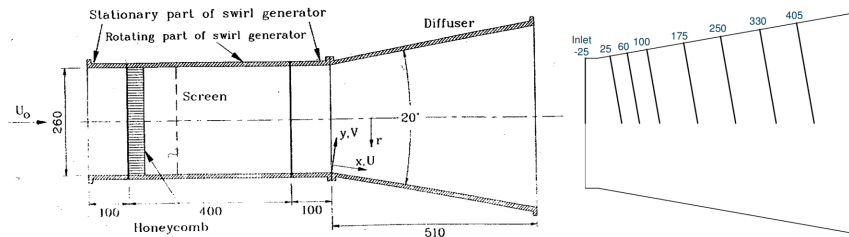


Figure: Experimental setup (Clausen *et al.*, (1993))

Validation case: ERCOFTAC conical diffuser

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 - Total divergence angle: 20°
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- Swirl sufficient to avoid boundary layer **separation**
- Swirl insufficient to cause **recirculating core** flow

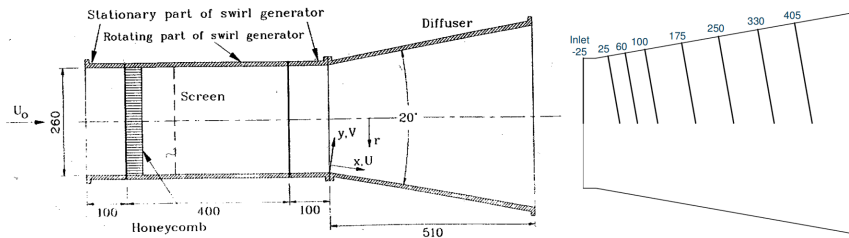


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Meshing and boundary conditions

Validation case: ERCOFTAC conical diffuser

- Meshes:

- Two-dimensional axisymmetric and three-dimensional meshes
- Outlet extension added: 10 inlet diameters long
- High-Re turbulence models: $30 < y^+ < 100$
- Low-Re turbulence models: $y^+ < 5$

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 - Inlet: measurements of Clausen *et al.* (1993)
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 - Walls: no-slip condition
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Turbulence models

Validation case: ERCOFTAC conical diffuser

Tested six turbulence models:

- High-Re models with wall functions:
 - Standard $k-\varepsilon$ (SKE)
 - Realisable $k-\varepsilon$ (RKE)
 - SST $k-\omega$ (SST)
 - $\overline{v'^2}-f$ (V2F)
- Low-Re models with integrated boundary layers:
 - SST $k-\omega$ (SSTLR)
 - $\overline{v'^2}-f$ (V2FLR)

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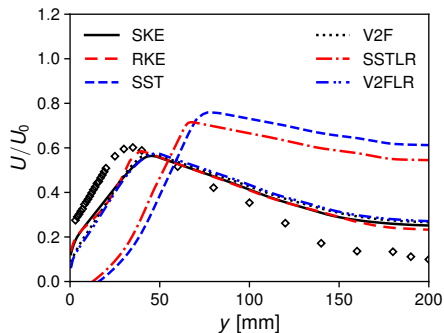


Figure: Streamwise velocity

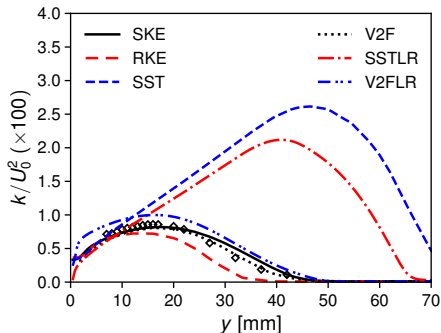


Figure: Turbulent kinetic energy

2D axisymmetric versus 3D simulations

Validation case: ERCOFTAC conical diffuser

- Test axisymmetric assumption
- Used standard $k-\varepsilon$ model

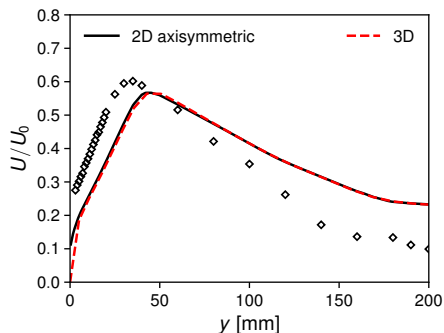


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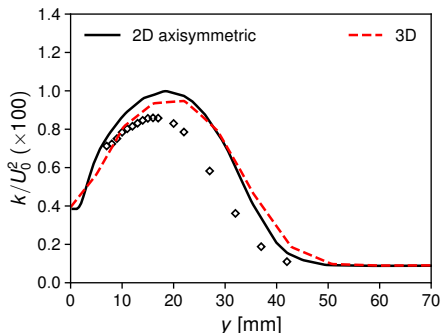


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Full-scale M-fan simulations

- Realisable $k-\epsilon$ model with wall functions
- Two-dimensional axisymmetric mesh

Full-scale M-fan simulations

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- Seven configurations to test:
 - 1 Outlet guide vanes
 - 2 Conical diffuser
 - 3 Conical diffuser with guide vanes at its inlet
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Full-scale M-fan simulations

- A guide vane with nine blades was designed
- Numerically modelled with the actuator disc model

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Full-scale M-fan simulations

- A guide vane with nine blades was designed
- Numerically modelled with the actuator disc model
- Pressure recovered: **15.9 Pa** ($K_{rec} = 0.37$)
- **13.6 %** of fan pressure rise at design flow rate

Case 2: Conical diffuser

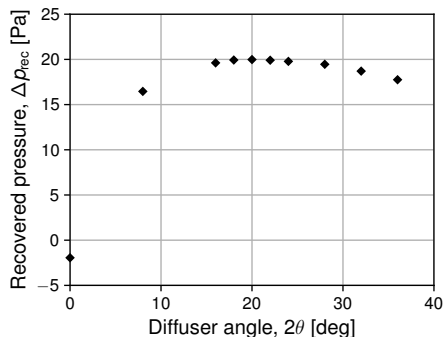
Full-scale M-fan simulations

- Diffuser length set equal to fan diameter
- Tested different divergence angles to find best diffuser performance

Case 2: Conical diffuser

Full-scale M-fan simulations

- Diffuser length set equal to fan diameter
- Tested different divergence angles to find best diffuser performance
- Best angle: $2\theta = 20^\circ$
- Pressure recovered: **20.0 Pa**
($K_{\text{rec}} = 0.46$)
- **17.1 %** of fan pressure rise at design flow rate



Conclusions

- Pressure recovery term:

$$K_{\text{rec}} = \alpha_{eF} - \alpha_{e7} (A_{\text{FC}}/A_7)^2 - K_{\text{dif}}$$

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 - Realisable $k-\varepsilon$ with wall functions

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- Validation study:
 - Two-dimensional axisymmetric simulations
 - Realisable $k-\varepsilon$ with wall functions
- M-fan simulations:
 - Swirl removal: 13.6 % pressure increase
 - Conical diffuser with 20° divergence angle: 17.1 % pressure increase